# **European Aviation Safety Agency**



# TYPE CERTIFICATE

# **EASA.A.583**

This Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EC) No. 748/2012 to

# COSTRUZIONI AERONAUTICHE TECNAM S.R.L.

VIA TASSO 478 80127 NAPOLI NA ITALY

and certifies that the product type design listed below complies with the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified on the associated:

Type Certificate Data Sheet number: EASA.A.583

Type Design - Model:

Date of Issue:

P2008 JC

27 September 2013

For the European Aviation Safety Agency,

Date of issue: 27 September 2013

Alain LEROY
Head of Products Department



# European Aviation Safety Agency

# **EASA**

# TYPE-CERTIFICATE DATA SHEET

**EASA.A.583** 

P2008 JC

Costruzioni Aeronautiche TECNAM S.r.I.

Via Tasso, 478 80127 Napoli ITALIA

Issue 01: 27 September 2013 TC issue

Issue 02: 24 July 2014 S/N 1001 is excluded from the TCDS

Issue 03: 23 April 2015 Added Hoffmann propeller

Added increment of weight

Updated Hoffmann TC reference

Changed 8.3 (coolant type)

Changed notes 1,2 and 3

Issue 04: 23 October 2015

Issue 06: 18 January 2016

Issue 05: 11 December 2015

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- II. Type Certificate Holder Record
- III. Change Record

#### **SECTION A:** P2008 JC

#### A.I. General

1. Data Sheet No.: **EASA.A.583** 

2. a) Type: P2008 JC

3. Airworthiness Category: CS-VLA Normal category

4. Type Certificate Holder: Costruzioni Aeronautiche Tecnam S.R.L.

> Via Tasso, 478 80127 Napoli

**ITALIA** 

5. Manufacturer: Costruzioni Aeronautiche Tecnam S.R.L.

> Via Tasso, 478 80127 Napoli

**ITALIA** 

6. Certification Application

Date:

09 May 2011

7. (Reserved) National

Certifying Authority

N/A

N/A

8. (Reserved) National

Authority Type Certificate

Date:

### A.II. <u>EASA Certification Basis</u>

1. Reference Date for

determining the applicable

requirements:

09 May 2011

2. Airworthiness Requirements: EASA CS-VLA amdt.1 dated 5 May 2009

CRI O-101 (SC-VFR night); CRI F-101 (SC-F-3 Special Conditions:

1309-01 Protection from the Effect of HIRF)

3. Exemptions: None

4. Deviations: None

5. Equivalent Safety Findings: None

6. Requirements elected to

comply:

None

7. Environmental Standards: CS-36 Amdt. 2 dated 31 August 2009, subpart C

with reference to ICAO Annex 16, Volume 1,

Chapter 10, Amdt. 9 dated 30 July 2009.

8. (Reserved) Additional

National Requirements:

N/A

9. (Reserved) N/A

# A.III. <u>Technical Characteristics and Operational Limitations</u>

1. Type Design Definition: Document no. 2008/008 "Type Design Definition"

2. Description: Single-engine, fixed pitch propeller, two seats, high

wing aeroplane equipped with fixed tricycle landing gear, featuring composite, aluminium and steel

construction.

3. Equipment: Equipment list, AFM, doc. No. 2008/100, Section 6

4. Dimensions:

Span 9,00 m (29.5 ft) Length 6,97 m (22.9 ft) Height 2,67 m (8.8 ft)

Wing Area 12,16 m<sup>2</sup> (130.9 ft<sup>2</sup>)

5. Engine:

5.1.1 Model: No.1 Bombardier-Rotax GmbH 912 S2

5.1.2 Type Certificate: EASA Type Certificate No. EASA.E.121

5.1.3 Limitations: Take-Off Power 73,5 kW (98.6 HP) at 5800 RPM

(5 minutes maximum)

Max continuous power 69 kW (92.5 HP) at 5500 RPM

Other engine's limitations are listed in doc. No.

2008/100 "P2008 JC Aircraft Flight Manual", Section 2

6. Load factors:

6.1 Basic: Flap UP Flap DOWN

Positive +4,0 g +2,0 g Negative -2,0 g 0,0 g

6.2 Optional (see Notes 2,3): Flap UP Flap DOWN

Positive +3,8 g +1,9 g Negative -1,9 g 0,0 g

7. Propeller:

7.1 Model: GT propellers: GT-2/173/VRR-FW101 SRTC

Type Certificate: EASA Type Certificate No. EASA.P.108

Number of blades: 2

Diameter: 1,730 m (68 in) – No reduction is permitted

Sense of Rotation: Clockwise (pilot's view)

7.2 Model (see Notes 1,3): Hoffmann KG: HO17GHM A 174 177C

Type Certificate: LBA Type Certificate No. 32.110/1 (EASA Approved)

Number of blades: 2

Diameter: 1,740 m (68,5 in) – No reduction is permitted

Sense of Rotation: Clockwise (pilot's view)

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8.1 Fuel: - MOGAS:

ASTM D4814 (min RON 95/AKI 91)

• EN 228 Super/Super plus (min. RON 95/AKI 91)

- AVGAS 100 LL (ASTM D910)

8.2 Oil: Only oil with API classification "SG" or higher.

Recommended by Rotax:

SHELL AeroShell Sport Plus 4API SL

Refer to Rotax SI-912-016 R4 for list of alternative recommended commercial brands and types.

8.3 Coolant: According to Aircraft Flight Manual

9. Fluid capacities:

9.1 Fuel: 2 Tanks: 62 litres each (16.38 US gallons)

Total: 124 litres (32.76 US gallons)
Usable: 120 litres (32 US gallons)

9.2 Oil: Total: 3 litres

Minimum: 2,5 litres

9.3 Coolant system Expansion tank: 0,25 litres

capacity: Overflow bottle: 0,5 litres

10. Air Speeds:

10.1 Basic: Never exceed speed V<sub>NE</sub> 141 KCAS

Maximum Structural Cruising Speed V<sub>NO</sub> 111 KCAS

Design Manoeuvring speed V<sub>A</sub> 98 KCAS

Operating Manoeuvring speed Vo 98 KCAS

Maximum flaps extended speed V<sub>FE</sub> 72 KCAS

10.2 Optional (see Notes 2,3): Never exceed speed V<sub>NE</sub> 139 KCAS

Maximum Structural Cruising Speed V<sub>NO</sub> 110 KCAS

Design Manoeuvring speed V<sub>A</sub> 97 KCAS

Operating Manoeuvring speed V<sub>0</sub> 97 KCAS

Maximum flaps extended speed V<sub>FE</sub> 71 KCAS

11. Maximum Operating

Altitude:

13,000 ft

12. All-weather Operations

Capability:

Day-VFR;

Night VFR is allowed on aeroplanes with KIT P/N 28-

13-1000-000 installed and operative.

Refer to KOEL contained in the AFM, doc. No.

2008/100, Section 2.

Flight into expected or actual icing conditions is

prohibited

13. Maximum Weights:

13.1 Basic: Max Take-Off: 630 kg (1388 lb)

> Max Landing: 630 kg (1388 lb)

Max Take-Off: 650 kg (1433 lb) 13.2 Optional (see Notes 2,3):

> Max Landing: 650 kg (1433 lb)

14. Centre of Gravity Forward Limit: 1,841 m (20% MAC) behind datum

Range: Aft Limit: 1,978 m (30% MAC) behind datum Mean Aerodynamic Chord is 1,373 m (54 in)

15. Datum:

Propeller support flange without spacer

Stabilator: 15°±2° to pitch up / 4°±2° to pitch down 16. Control surface deflections:

Stabilator Trim Tab: 12 ±1° downward / 2°±1° upward

Aileron: 22°±2° upward / 14°±2° downward

Rudder: 25°±2° left / 25°±2° right

Flaps: 0° Fully Retracted / 35°±1° Fully Extended

seat track supporting beams (see procedure in doc. 17. Levelling Means:

No. 2008/100 "P2008 JC Aircraft Flight Manual",

Section 6)

18. Minimum Flight Crew: 1

19. Maximum Passenger

Seating Capacity:

Max Allowable Load: 20 kg (44 lb) 20. Baggage/Cargo

1

Compartments: Location:2,42 m (95.28 in) from datum

Nose Wheel Tyre Size: 5.00-5, Type III 21. Wheels and Tyres:

> 5.00-5, Type III Main Wheel Tyre Size

For approved Types and rating see AMM, doc No.

2008/101

1002 to subsequent 22. Serial Numbers Eligible:

# A.IV. Operating and Service Instructions

1. Flight Manual: Doc. No. 2008/100 "P2008JC Aircraft Flight

Manual" Last issue

2. Technical Manual: Doc. No. 2008/101 "P2008JC Aircraft Maintenance

Manual" Last issue

3. Spare Parts Catalogue: Doc. No. 2008/102 "P2008JC Illustrated Parts

Catalogue" Last issue

4. Instruments and aggregates: Doc. No. 2008/101 "P2008JC Aircraft Maintenance

Manual" Last issue

### A.V. Notes:

- 1) When MOD 2008/029 (EASA approval 10052448) or MOD 2008/045 (EASA approval 10056252) is installed
- 2) When MOD 2008/027 (EASA approval 10053015) or MOD 2008/045 (EASA approval 10056252) is installed
- 3) MOD description:
  - MOD2008/027: MTOW increment to 650kg
  - MOD2008/029: Hoffmann propeller
  - MOD2008/045: Hoffmann propeller combined with MTOW increment to 650kg

# **ADMINISTRATIVE SECTION**

### I. Acronyms

AFM – Aircraft Flight Manual

AMM - Aircraft Maintenance Manual

API – American Petroleum Industry

ASTM - American Society for Testing and Materials

CRI - Certification Review Item

CS - Certification Specification

VLA – Very Light Aircraft

EASA – European Aviation Safety Agency

ICAO – International Civil Aviation Organization

IPC - Illustrated Part Catalogue

KCAS – Knots Calibrated Air Speed

**KOEL – Kind of Operations Equipment List** 

MAC - Mean Aerodynamic Chord

MLW - Maximum Landing Weight

MTOW - Maximum Take-Off Weight

MZFW - Maximum Zero Fuel Weight

TC – Type Certificate

TCDS - Type Certificate Data Sheet

VFR - Visual Flight Rules

### II. Type Certificate Holder Record

TC Holder	Period
Costruzioni Aeronautiche TECNAM S.r.l.	Effective
Via Tasso, 478	
80127 Napoli	
ITALIA	

### III. Change Record

Issue	Date	Changes	TC Issue No. & Date
Issue 01	27 September 2013	Initial Issue	Is.01, 27 Sep 2013
Issue 02	24 July 2014	S/N 1001 is excluded from the TCDS	
Issue 03	23 April 2015	Increment of weight (mod 2008/027) and new propeller (MOD 2008/029) are added	
Issue 04	23 October 2015	Updated TC Hoffmann reference	
Issue 05	11 December 2015	Changed 8.3 (coolant type)	
Issue 06	18 January 2016	Changed notes 1, 2 and 3	